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Evaluation of buckling characteristics of thin cylindrical shells with cutouts for aerospace applications

Raj Pranav

rajpranav4040@gmail.com

Malla Reddy College of Engineering and Technology,
Secunderabad, Telangana,

Ashok Kanna

sarvalaashok@gmail.com

Malla Reddy College of Engineering and Technology,
Secunderabad, Telangana,

Akshay Kumar

chintuakki111@gmail.com

Malla Reddy College of Engineering and Technology,
Secunderabad, Telangana,

Ajith Raj R.

ajitharaj@gmail.com

Malla Reddy College of Engineering and Technology,
Secunderabad, Telangana,

ABSTRACT

Aerospace industry finds a wide application of thin cylindrical shells. Cutouts in various structures are also an important application particularly in aircraft and aerospace industry which is a significant area of research today. The load carrying capacity of these structures is determined by the buckling characteristics of that specimen mainly in aircraft since the primary cause of failure is elastic buckling. Even though cut outs in these structures decreases the buckling strength due to the stress concentrations, it cannot be avoided in many engineering applications particularly in aircraft components. Cylindrical shells made of mild steel are considered for the study. In this paper the buckling behavior of the shells made without cut outs, with cutouts, and with semi cutouts of constant diameter are studied.

Keywords— Thin Cylindrical shells, Cutouts, Buckling behavior, Aerospace Applications

1. INTRODUCTION

Buckling of shells is one of the most important considerations in auxiliary designing. Small distortions happen in the shell when it retains strain energy. Notwithstanding, it misshapes altogether at the point when it consumes bending strain energy. In the event that the shell is loaded and arrives at a phase where this put away strain energy and changed over into an equal bending energy and the shell fails which is called buckling.

Shells with cutouts are utilized in various structures like Storage tanks, Turbine towers, Aircraft parts, Submarine structure, Pipelines, Chimneys, and so on. These structures are subjected to various loading conditions such as bending, torsion, axial compression etc. Buckling of such structures is a significant marvel in basic mechanics, in light of the fact that buckling frequently prompts failure of structures. The buckling quality of thin cylindrical shells under axial compression is increasingly delicate to defects than shells under other stacking conditions. Most of these structures have round cutouts which experiences elastic failure as the primary failure mode more particularly in case of aircraft components. Therefore, this area has become an important area of research in recent days. The buckling behavior of the cylindrical shells made without cut outs, with cutouts, and with semi cutouts of constant diameter is studied. Cylindrical shell made of mild steel is considered for this study.

2. LITERATURE REVIEW

Miladi et.al [1] conducted investigates on cylindrical shells made of steel with circular cut outs. They reported that the cut out reduces the load bearing capacity of thin cylindrical shells around 10 to 35 %. The examination likewise indicated that thinner shells are progressively powerless to the patterns contrasted with thicker ones. Anil Kumar Nagari et.al [2] led test examination of buckling of covered composite cylindrical shells with and without cutouts exposed to compression. The outcomes which were gotten in this analysis show how the defects like cutouts will influence the buckling conduct of the very thin shells. The load withstanding limit is low for shells with cutouts. Stasiewicz [3] examined the impact of circular cut outs on cylindrical shells made of steel, metal and aluminum alloys. This examination demonstrated that 14 mm cut outs bring about decrease in stress by 40%. Shariati et.al [4] focused on the buckling of steel cylindrical shells with a circular cut out. They found that cutouts reduce the load limit around 10 to 35%. The investigation additionally demonstrated that thin shells with cutouts are poor to the patterns

contrasted with thicker ones. L. Gangadhar et.al [5] contemplated buckling examination of thin walled cylindrical composite shells with and without cutouts by applying compressive load on shell. It was discovered that the buckling load is most extreme in round cutout and least in the rectangular cutout. Brogan et.al. [6] Concentrated on the buckling of aluminum chambers with square patterns. In the models it was discovered that presence of cut out brought about decrease of buckling load by 30%. Neethi. B et.al [7] led a systematic investigation on the buckling quality of a composite cylindrical shell with cutouts. This investigation uncovers the buckling quality of a composite cylindrical shell with cutouts scientifically.

3. EXPERIMENTAL INVESTIGATION

Cylindrical shells were made without cut outs, with cutouts, and with semi cutouts of constant diameter. Mild steel is used and welding is used for the manufacturing of cylindrical shells. The characteristics of buckling due to the effect of change in cutouts at different positions are also studied. For this purpose, cutouts were made at L/3, L/6 and L/9 distance from the shell's top and the change in buckling load was studied. The specimens used for the study are thin cylindrical shells of 500mm long 100mm diameter, and 1 mm thickness. Cutouts selected for this study is circular with 3mm diameter. Each specimen is subjected to axial compressive load using Universal Testing Machine. The load was increased till buckling failure occurs and the corresponding buckling load is found. The experimental setup used for compression test is shown in figure 1.



Fig. 1: Experimental Setup

3.1 Effect of Cutouts

Cylindrical shells without cut outs, with cutouts, and with semi cutouts of 3mm diameters are taken for the study. The results of the specimen in terms of buckling load are shown in table 1. It was found that the cylindrical shells without cutouts shows high buckling strength than the other two. The specimen with semi cutouts (cutouts in one half and without cutouts on the other half from the middle of the specimen) shows a less value in buckling strength but it is very high when compared with the specimen with full cutouts all over. Therefore, this experiment reveals that the specimen with semi cutouts can be a better option when cutouts cannot be avoided in case of special applications. A graphical representation of the result is shown in Figure 2. The stresses will concentrate over the regions of cutouts and cause buckling failure.

Table 1: Result of buckling load on different Specimen

Specimen	Without Cutouts	With Cutouts	With Semi Cutouts
Buckling Load (KN)	66.8	54.2	61.4

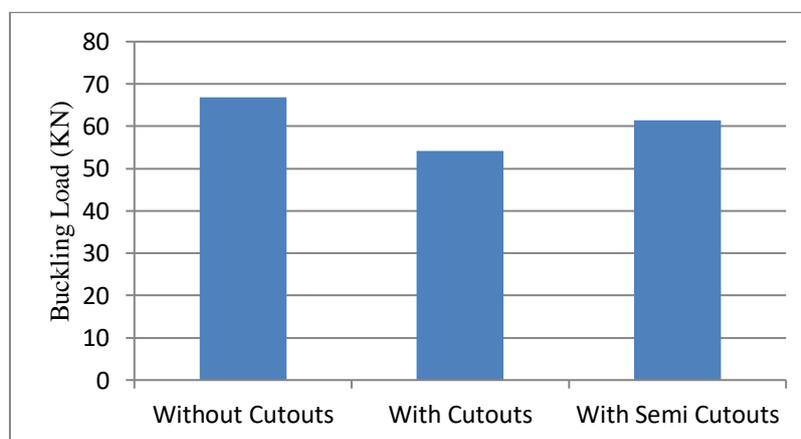


Fig. 2: Effect of cutouts on Buckling Load

4. CONCLUSION

Cut outs cannot be avoided in many engineering applications particularly in aircraft components. But the presence of cutouts will decrease the buckling strength of the material. Therefore, various studies have been conducted in this area to identify the shape,

position etc. to optimize the cutouts so that the buckling behaviour is improved. In this study the buckling behavior of the shells made without cut outs, with cutouts, and with semi cutouts of constant diameter are studied. The specimen with semi cutouts shows a less value in buckling strength when compared with specimen without cutouts but shows a very high value when compared with the specimen with cutouts. Therefore, this current experiment reveals that the specimen with semi cutouts can be a better option to use when cutouts cannot be avoided in case of special applications. Further research has to be carried out to optimize the shape, position etc. of cutouts.

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